

**RANS MODEL DEVELOPMENT
IN SUPERSONIC BOUNDARY LAYER FLOW WITH AND
WITHOUT SHOCKS**

**DÉVELOPPEMENT D'UN MODÈLE RANS
EN COUCHE LIMITE D'ÉCOULEMENT SUPERSONIQUE
AVEC ET SANS CHOCS**

T. B. Gatski

Département Fluides, Thermique, Combustion

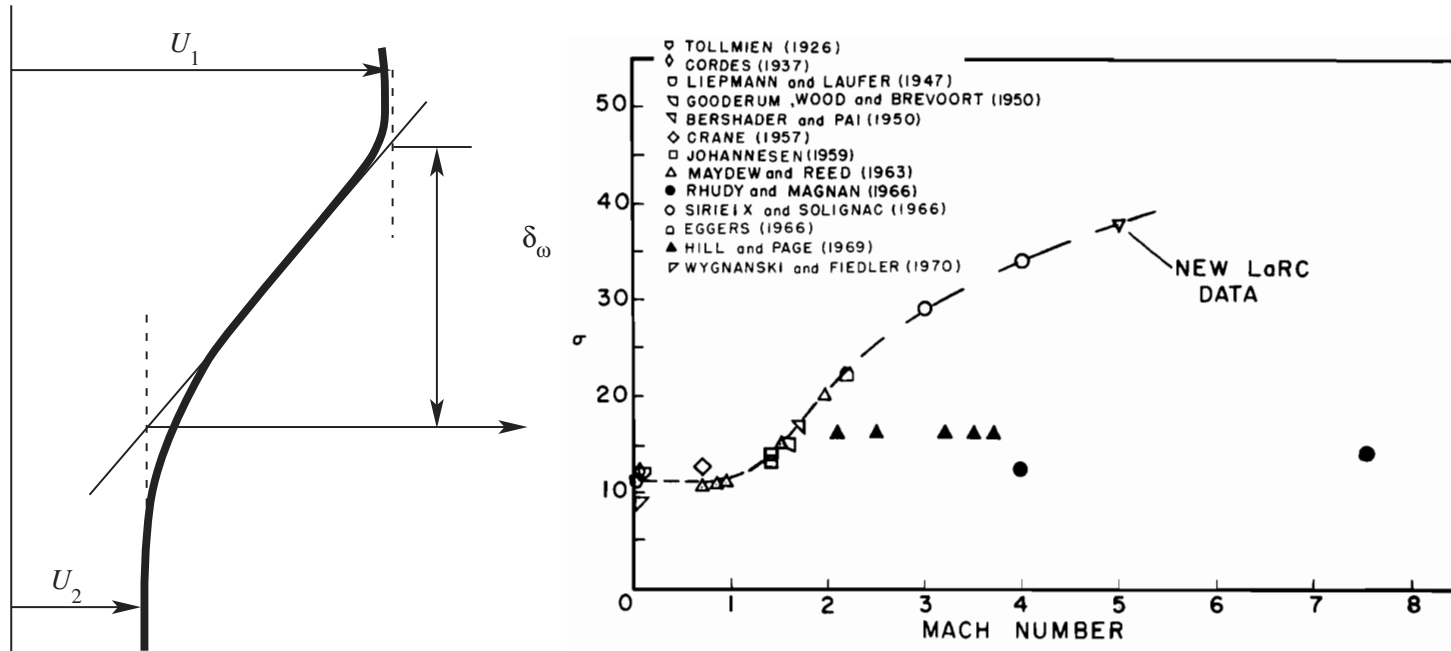
Institut PPRIME (UPR 3346) CNRS, Université de Poitiers, ENSMA

F86962 Futuroscope Chasseneuil Cedex, France

SOME BACKGROUND

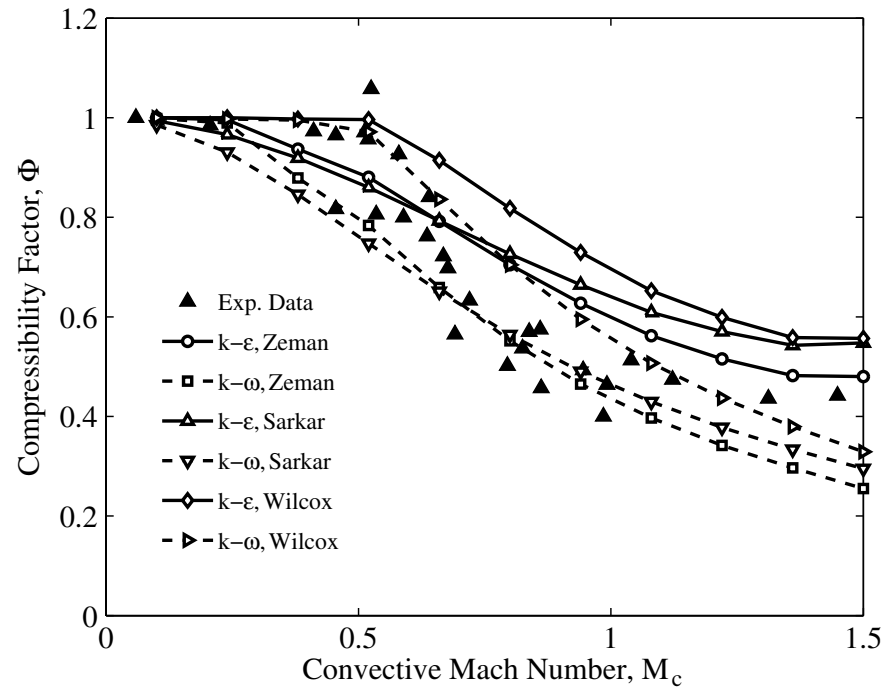
- RANS: Most commonly used prediction method
 - Variable density extensions incompressible forms
- Compressibility models
 - Dilatation models
 - Influence and role of pressure fluctuations
- Reduced mixing-layer spreading rate with convective Mach number motivated early research
 - Early studies assumed changes in density distribution important

SOME BACKGROUND



Schematic of self-similar mixing layer longitudinal mean velocity profile, and variation of spreading rate with Mach number $U_2 = 0$: \bullet , values for incompressible flow, and other symbols for compressible mixing layers. (Birch and Eggers, 1972)

SOME BACKGROUND



$$\left(\frac{d\delta}{dx} \right)_c = \Phi \left(\frac{d\delta}{dx} \right)_i$$

Growth rate from different experimental mixing layer studies selected by Barone et al. (2006).

OUTLINE

Data from DNS of ZPG Supersonic BL Flow, $M_\infty = 2.25$

- Without Shock
 - Equilibrium modeling considerations
 - Mean velocity, thermodynamic variable distributions across BL
 - Turbulent velocity, thermodynamic and scalar flux correlation distributions across BL
 - Kinetic energy, energy dissipation rate, temperature variance and scalar flux budgets across BL
- With shock
 - Non-equilibrium modeling considerations
 - Strength of impinging shock/flow separation
 - Streamwise relaxation effects

PHYSICAL PARAMETERS

Adiabatic

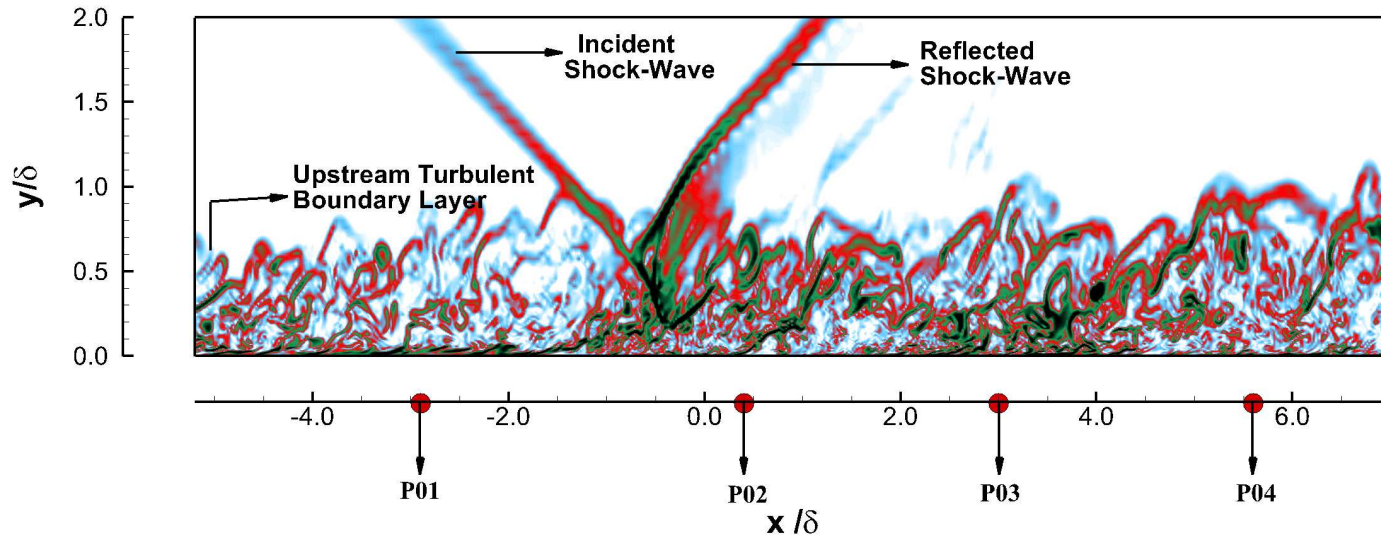
- Freestream Mach Number
 $M_\infty = 2.25$
- Freestream unit Reynolds number
 $Re/m. = 25 \times 10^6$
- Momentum Thickness Reynolds number $Re_\theta = 3706$
- Shape factor $H = 3.40$
- $q_w = 0, T_{aw} = 342K, T_r = 323K,$
 $T_\infty = 170K, r = 0.89$
- $(-B_q, M_\tau) = (0.0, 0.077)$

Isothermal

- Freestream Mach Number
 $M_\infty = 2.25$
- Freestream unit Reynolds number
 $Re/m. = 25 \times 10^6$
- Momentum Thickness Reynolds number $Re_\theta = 3798$
- Shape factor $H = 2.65$
- $T_w = 230K, T_{aw} = 342K,$
 $T_w/T_{aw} = 0.67, T_\infty = 170K$
- $(-B_q, M_\tau) = (0.017, 0.079)$

$$B_q = \frac{q_w}{\bar{\rho}_w c_p u_\tau \tilde{T}_w}, \quad M_\tau = \frac{u_\tau}{c_w}$$

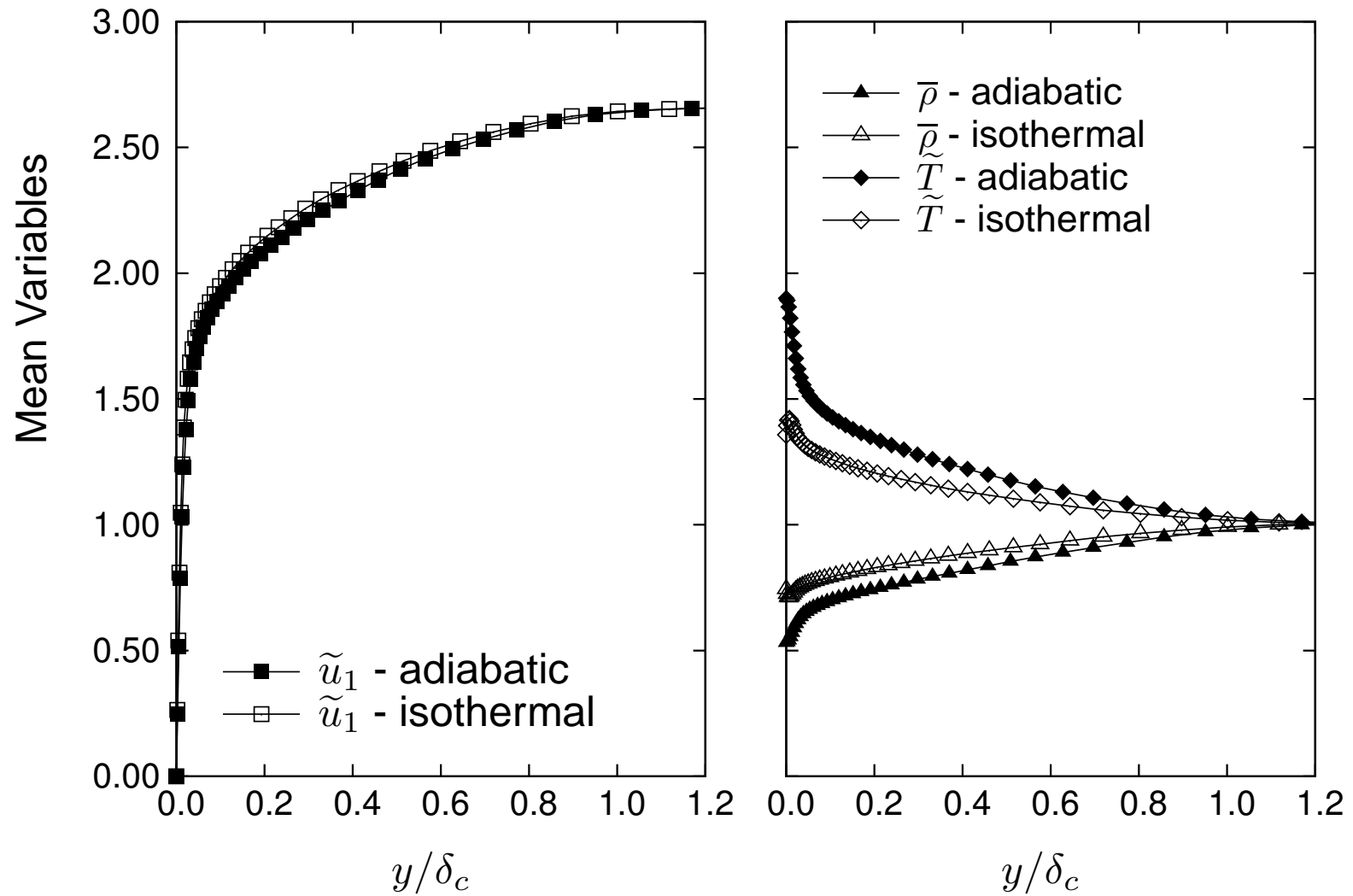
SHOCK CASE: DIAGNOSTIC POSITIONS



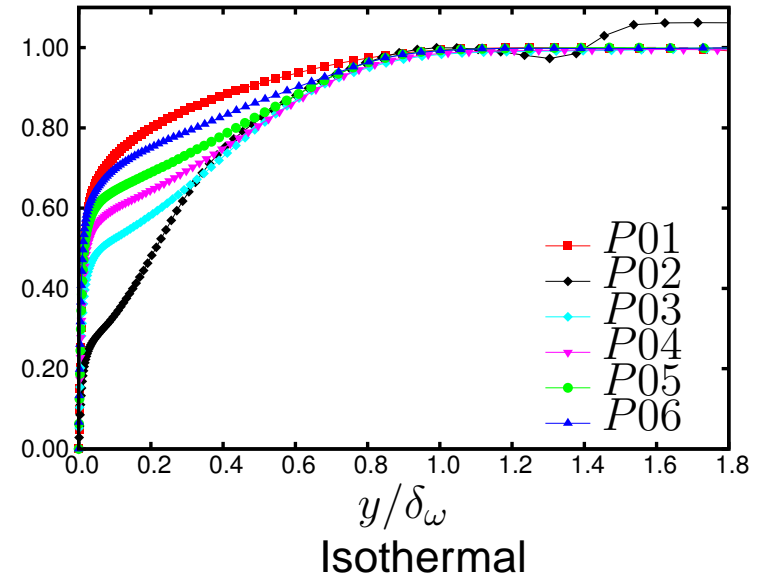
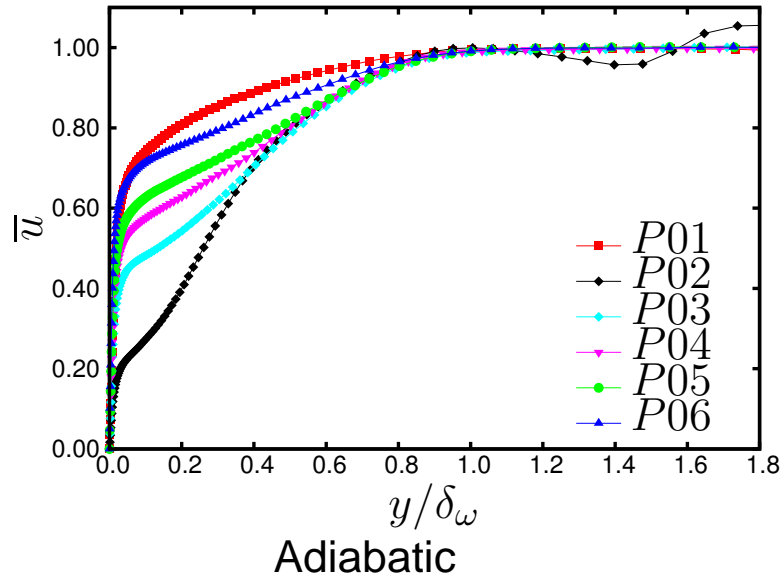
Diagnostic plane positions (x_i , impingement position of incident shock-wave; δ_c^o , boundary layer thickness at station $P01$).

<i>Plane</i>	$x_{pos} = (x - x_i) / \delta_c^o$	
	Adiabatic	Isothermal
$P01$	-2.9	-2.9
$P02$	0.4	0.4
$P03$	3.0	3.0
$P04$	5.6	5.6
$P05$	8.2	8.2
$P06$	16.7	16.7

MEAN FIELD



MEAN VELOCITY RELAXATION



WALL AND SEMI-LOCAL SCALING

- Proper choice of scaling dictated by collapse of correlation data
 - Usual viscous normalization

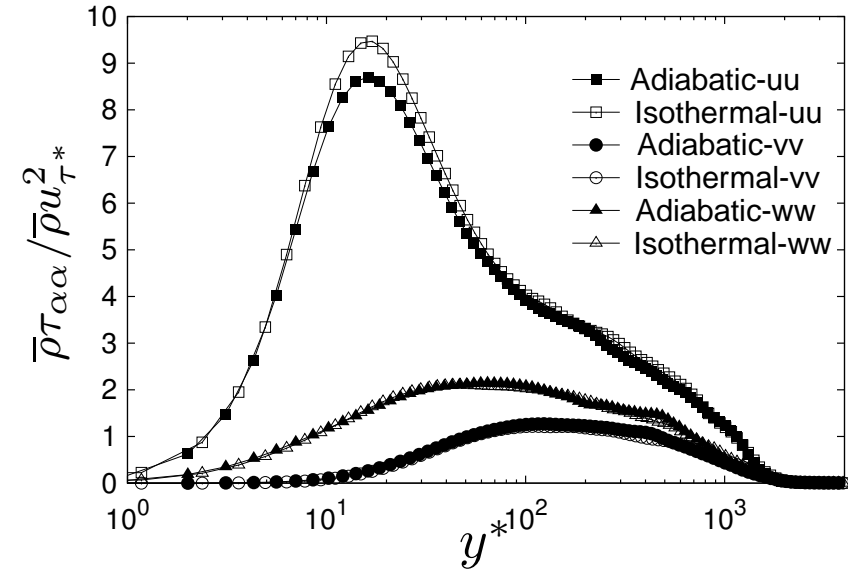
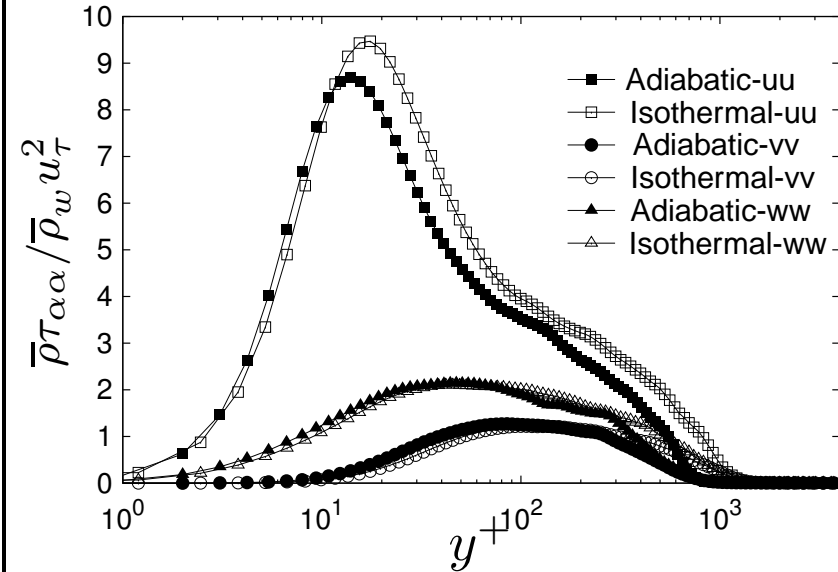
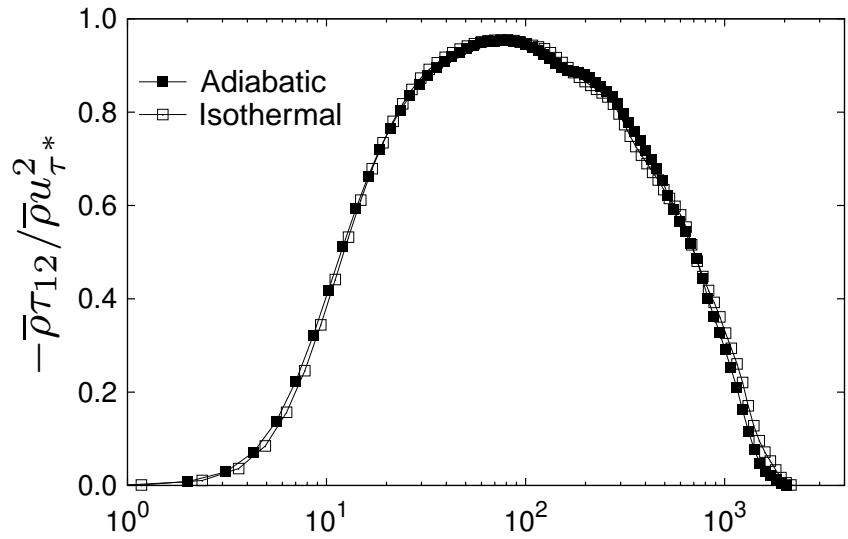
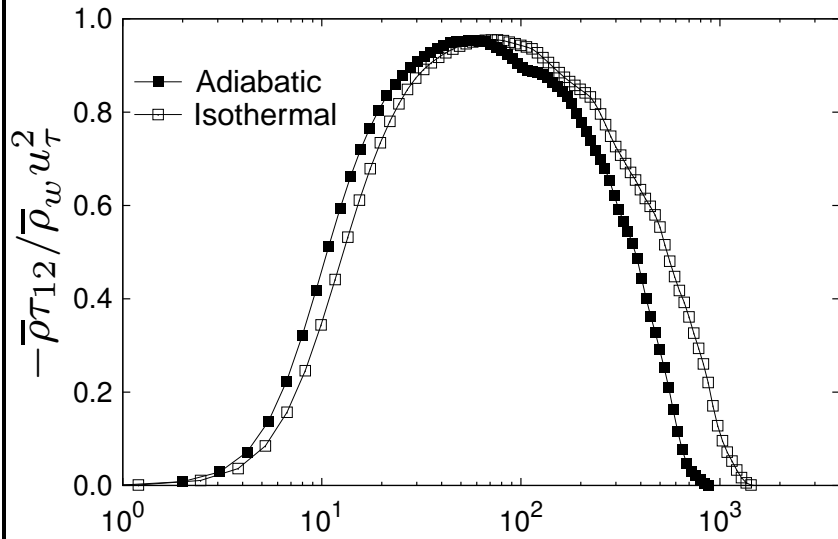
$$u_\tau = \sqrt{\tau_w / \bar{\rho}_w}, \quad \bar{\nu}_w = \bar{\mu}_w / \bar{\rho}_w$$

- Semi-local viscous normalization

$$u_{\tau^*} = \sqrt{\tau_w / \bar{\rho}}, \quad \bar{\nu} = \bar{\mu} / \bar{\rho}$$

- Need to account for density variations across boundary layer

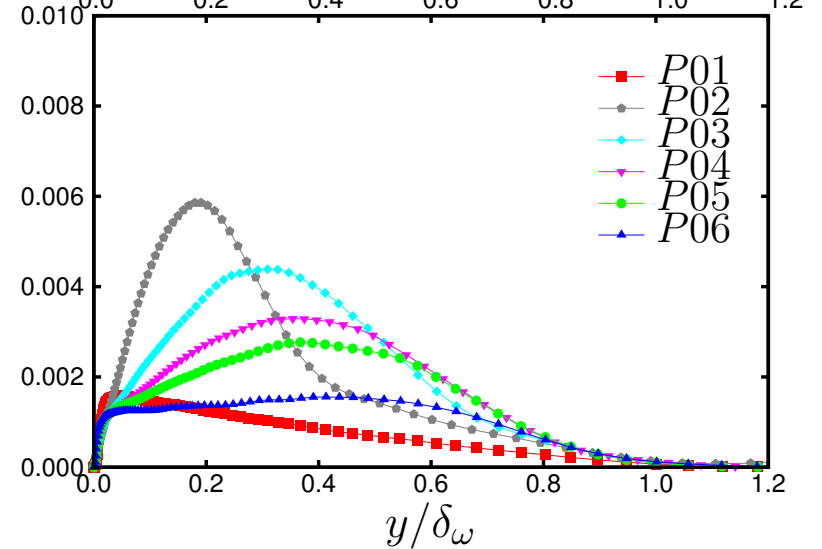
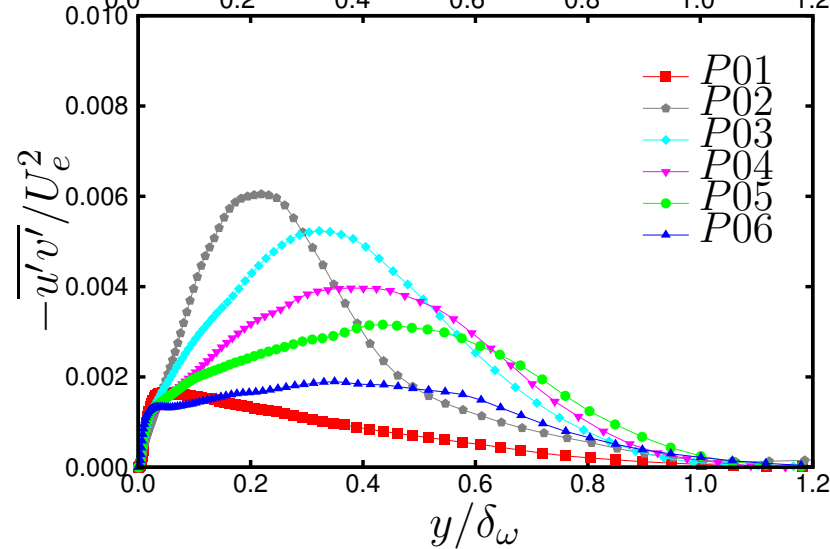
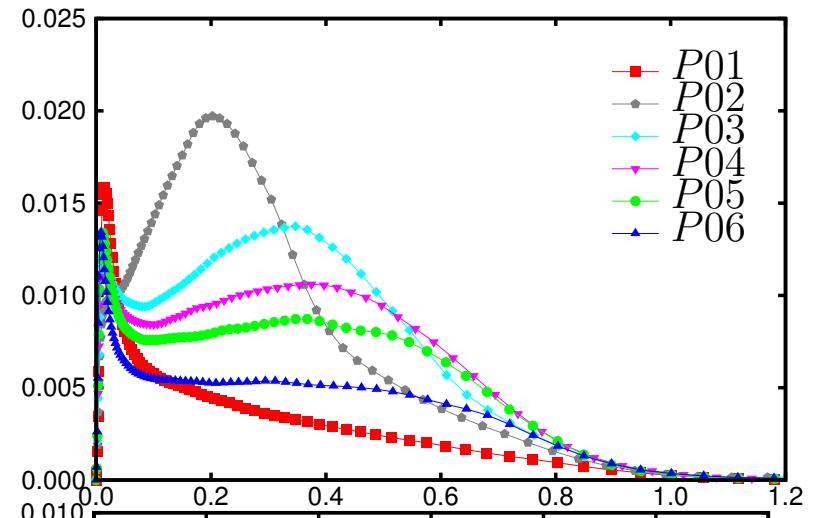
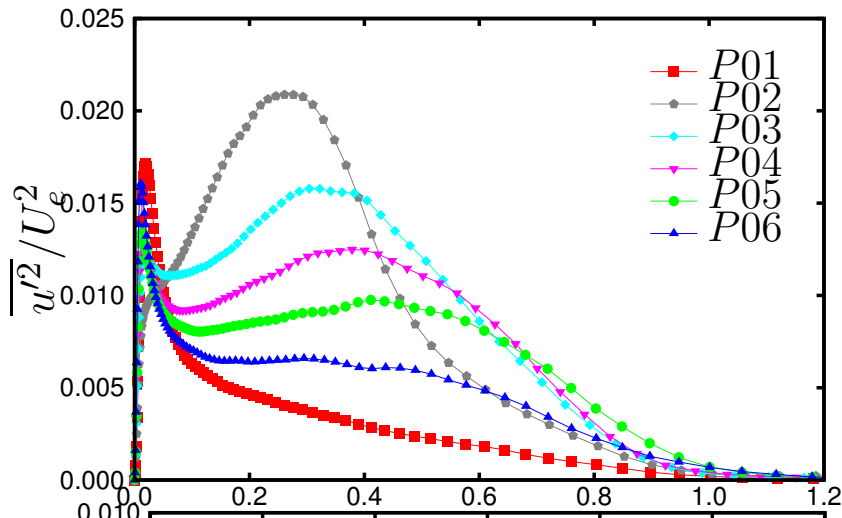
WALL AND SEMI-LOCAL SCALING



REYNOLDS STRESS DOWNSTREAM RELAXATION

Adiabatic

Isothermal



Turbulent Reynolds stress distributions across boundary layer

TURBULENT KINETIC ENERGY BUDGET

$$-A_K + P_K + \Pi_K + T_K + D_K - \epsilon + M_K = 0 ,$$

$$A_K = \bar{\rho} \tilde{u}_j \left(\frac{\partial K}{\partial x_j} \right) \text{ Advection (negligibly small)}$$

$$P_K = -\bar{\rho} \tau_{ik} \tilde{S}_{ki} \text{ Production}$$

$$\Pi_K = \overline{p' s'_{kk}} \text{ Pressure-dilatation}$$

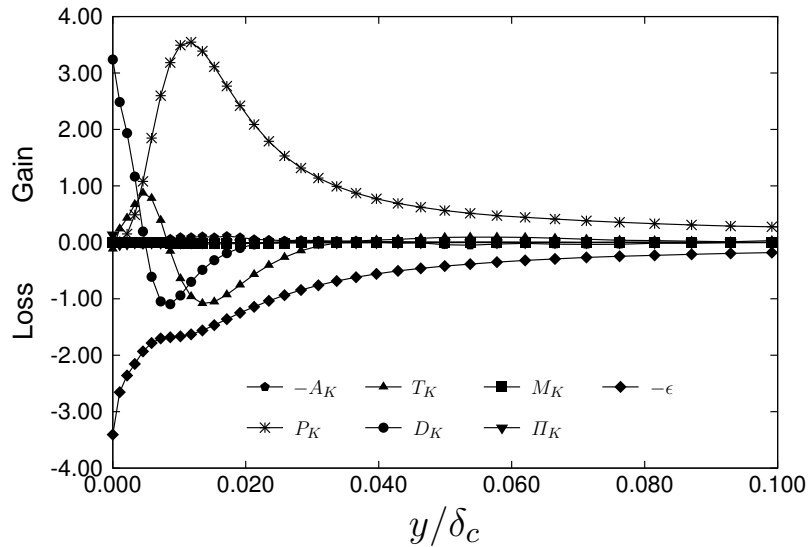
$$T_K = -\frac{\partial}{\partial x_k} \left[\frac{\overline{\rho u''_i u''_i u''_k}}{2} + \overline{p' u'_i} \delta_{ik} \right] \text{ Turbulent transport}$$

$$D_K = -\frac{\partial}{\partial x_k} \left[\overline{\sigma'_{ik} u'_i} \right] \text{ Viscous diffusion}$$

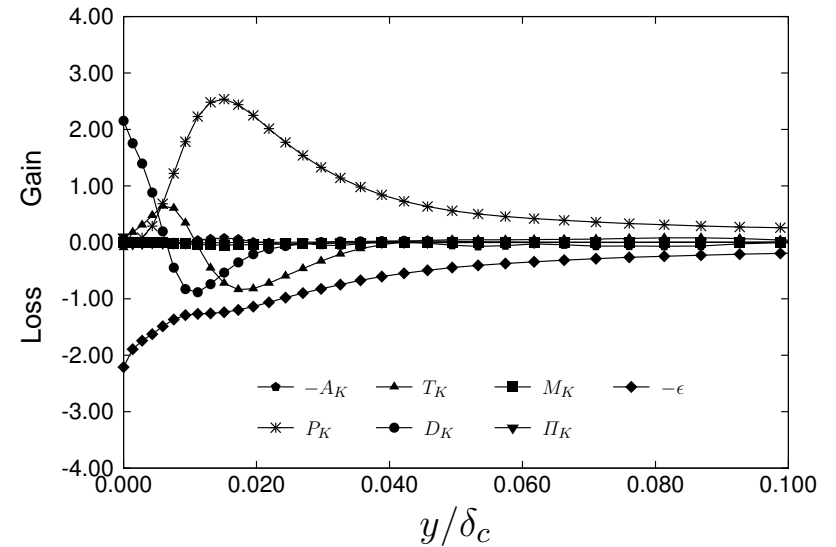
$$\epsilon = \overline{\sigma'_{ik} s'_{ki}} \text{ Dissipation}$$

$$M_K = \overline{\rho' u'_i} \left(\frac{\partial \bar{p}}{\partial x_i} - \frac{\partial \bar{\sigma}_{ik}}{\partial x_k} \right) \text{ Mass flux}$$

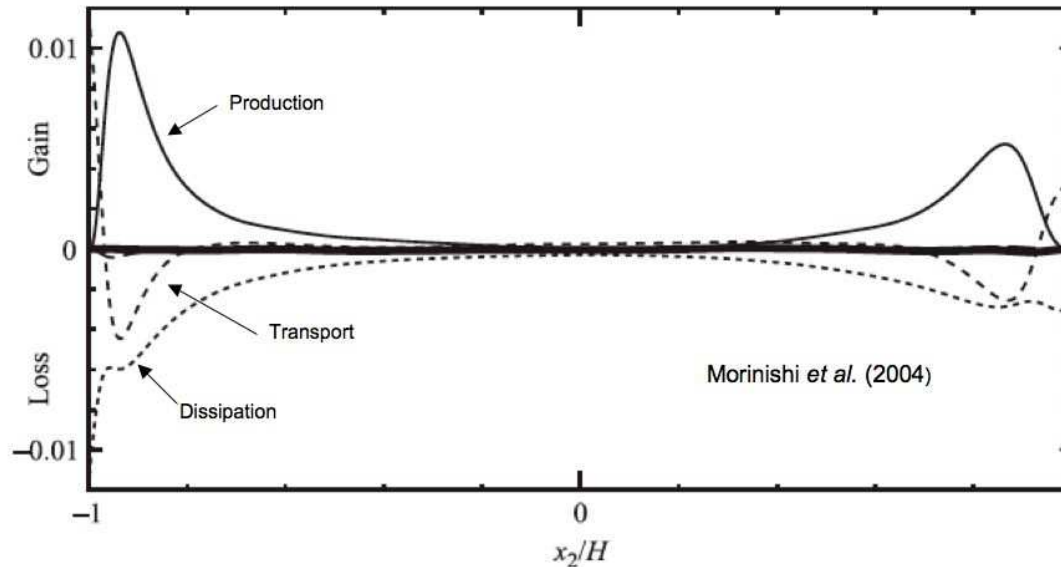
TURBULENT KINETIC ENERGY BUDGET



Isothermal



Adiabatic



Morinishi et al. (2004)

TURBULENT STRESS ANISOTROPY EQUATION

$$\begin{aligned}
 \frac{Db_{ij}}{Dt} - \frac{D_K}{K} \left[d_{ij}^{(t-\mu)} - b_{ij} \right] &= - \left[\frac{P_K}{\epsilon} + \frac{\Pi_K}{\epsilon} + \frac{M_K}{\epsilon} - 1 \right] \frac{b_{ij}}{\tau} + \frac{d_{ij}^{(\epsilon)}}{\tau} \\
 &+ \frac{1}{2K} \left[\left(\Pi_{ij} - \frac{2\delta_{ij}}{3} \Pi_K \right) + \left(M_{ij} - \frac{2\delta_{ij}}{3} M_K \right) \right] \\
 &- \frac{2}{3} \left(\tilde{S}_{ij} - \frac{\delta_{ij}}{3} \tilde{S}_{ii} \right) + \left(b_{ik} \tilde{W}_{kj} - \tilde{W}_{ik} b_{kj} \right) \\
 &- \left(b_{ik} \tilde{S}_{kj} + \tilde{S}_{ik} b_{kj} - \frac{2}{3} \left[\mathbf{b} : \tilde{\mathbf{S}} \right] \delta_{ij} \right)
 \end{aligned}$$

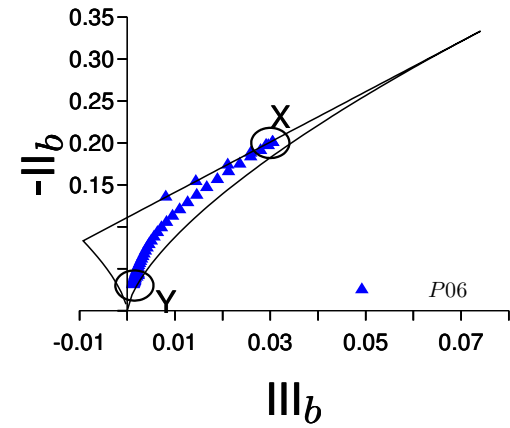
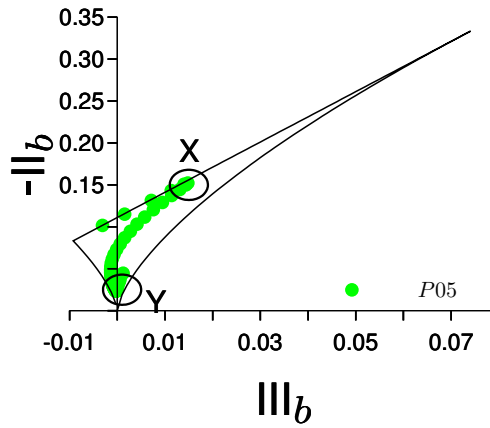
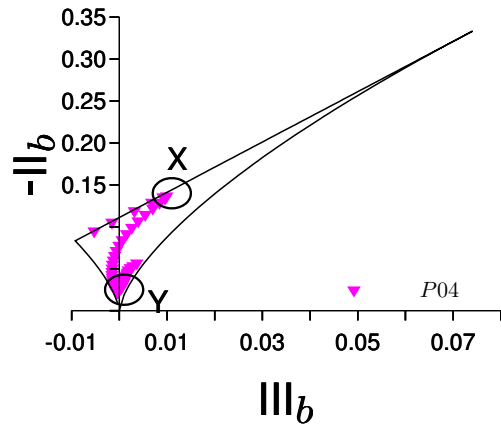
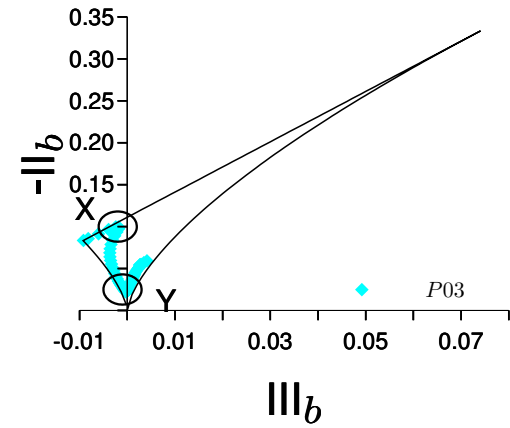
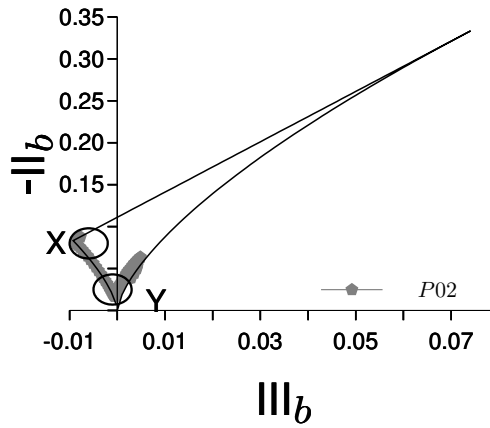
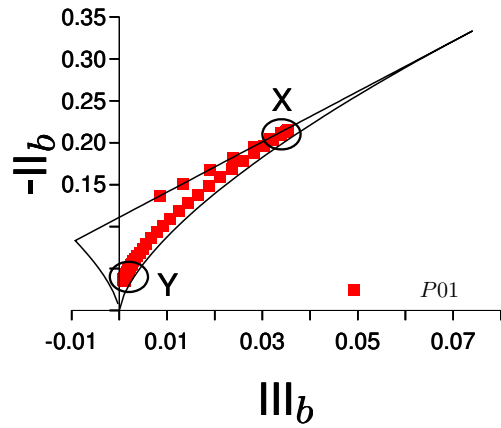
where

$$d_{ij}^{(t-\mu)} = \frac{D_{ij}}{2D_K} - \frac{\delta_{ij}}{3}, \quad \tau = K/\epsilon, \quad d_{ij}^{(\epsilon)} = \frac{\epsilon_{ij}}{2\epsilon} - \frac{\delta_{ij}}{3}$$

TURBULENT STRESS ANISOTROPY EQUATION

$$\begin{aligned}
 \frac{Db_{ij}}{Dt} - \frac{DK}{K} \left[d_{ij}^{(t-\mu)} - b_{ij} \right] &= - \left[\frac{PK}{K} + \frac{\Pi_K}{K} + \frac{MK}{K} \right] b_{ij} + \frac{\epsilon}{K} \left[d_{ij}^{(\epsilon)} + b_{ij} \right] \\
 &+ \frac{\Pi_K}{K} d_{ij}^{(\Pi)} + \frac{MK}{K} d_{ij}^{(M)} \\
 &- \frac{2}{3} \left(\tilde{S}_{ij} - \frac{\delta_{ij}}{3} \tilde{S}_{ii} \right) + \left(b_{ik} \tilde{W}_{kj} - \tilde{W}_{ik} b_{kj} \right) \\
 &- \left(b_{ik} \tilde{S}_{kj} + \tilde{S}_{ik} b_{kj} - \frac{2}{3} \left[\mathbf{b} : \tilde{\mathbf{S}} \right] \delta_{ij} \right)
 \end{aligned}$$

TURBULENT STRESS ANISOTROPY EQUATION



Adiabatic

PRESSURE CORRELATION TERMS

- Pressure-strain rate correlation

$$\begin{aligned}\bar{\rho}\Pi_{ij} &= \frac{2}{3}\overline{p' s'_{kk}}\delta_{ij} + 2\left(\overline{p' s'_{ij}} - \frac{\delta_{ij}}{3}\overline{p' s'_{kk}}\right) \\ &= \frac{2}{3}\bar{\rho}\Pi\delta_{ij} + \bar{\rho}\Pi_{ij}^d\end{aligned}$$

PRESSURE CORRELATION TERMS

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$$\begin{aligned}\bar{\rho}\Pi_{ij} &= \frac{2}{3}\overline{p' s'_{kk}}\delta_{ij} + 2\left(\overline{p' s'_{ij}} - \frac{\delta_{ij}}{3}\overline{p' s'_{kk}}\right) \\ &= \frac{2}{3}\bar{\rho}\Pi\delta_{ij} + \bar{\rho}\Pi_{ij}^d\end{aligned}$$

- Pressure-variance equation

$$\bar{\rho}\Pi = -\frac{1}{c^2}\frac{D}{Dt}\left(\frac{\overline{p'^2}}{2}\right) - \overline{p' u'_j}\frac{\partial\bar{\rho}}{\partial x_j} - \frac{\gamma\overline{p'^2}}{c^2}\tilde{S}_{jj} + \text{hot.}$$

$$\bar{\rho}\Pi = \begin{cases} \alpha_{Zd}\left(\frac{\overline{p'^2}}{p_e^2} - 1\right)\bar{\rho}\varepsilon M_t + \alpha_{Zc}\frac{\overline{p'^2}}{\gamma\bar{p}}\tilde{S}_{kk} & \text{Zeman (1991)} \\ \alpha_{Sd}\bar{\rho}\varepsilon M_t^2 + \alpha_{Sc}\gamma\bar{p}\tilde{S}_{kk}M_t^4 & \text{Sarkar (1992)} \end{cases}$$

PRESSURE CORRELATION TERMS

- Pressure-strain rate correlation

$$\begin{aligned}\bar{\rho}\Pi_{ij} &= \frac{2}{3}\overline{p' s'_{kk}}\delta_{ij} + 2\left(\overline{p' s'_{ij}} - \frac{\delta_{ij}}{3}\overline{p' s'_{kk}}\right) \\ &= \frac{2}{3}\bar{\rho}\Pi\delta_{ij} + \bar{\rho}\Pi_{ij}^d\end{aligned}$$

- Convective wave equation

$$\begin{aligned}&\left\{ M_t^2 \left[\frac{\partial}{\partial t} + \tilde{u}_j \left(\frac{M_g}{M_t} \right) \frac{\partial}{\partial x_j} \right]^2 - \frac{\partial^2}{\partial x_j \partial x_j} \right\} p' \\ &= 2 \left(\frac{M_g}{M_t} \right) \left\{ \left[\frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial (\rho u_j'')}{\partial x_i} \right] + \left(\frac{M_g}{M_t} \right) \rho' \left[\frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial \tilde{u}_j}{\partial x_i} \right] \right\} \\ &\quad + \frac{\partial^2}{\partial x_i \partial x_j} \left[\bar{\rho}(u_i'' u_j'' - \widetilde{u_i'' u_j''}) + \rho' u_i'' u_j'' \right],\end{aligned}$$

PRESSURE CORRELATION TERMS

- Pressure-strain rate correlation

$$\begin{aligned}\bar{\rho}\Pi_{ij} &= \frac{2}{3}\overline{p' s'_{kk}}\delta_{ij} + 2\left(\overline{p' s'_{ij}} - \frac{\delta_{ij}}{3}\overline{p' s'_{kk}}\right) \\ &= \frac{2}{3}\bar{\rho}\Pi\delta_{ij} + \bar{\rho}\Pi_{ij}^d\end{aligned}$$

- Poisson equation – time derivative of density in forcing term

$$\begin{aligned}-\frac{\partial^2 p'}{\partial x_i \partial x_i} &= -\frac{\partial^2}{\partial x_i \partial x_j} \left(\overline{\rho u_i'' u_j''} - \rho u_i'' u_j'' \right) \\ &\quad + 2\frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial(\rho u_j'')}{\partial x_i} + \rho' \frac{\partial \tilde{u}_i}{\partial x_j} \frac{\partial \tilde{u}_j}{\partial x_i} - \frac{D^2 \rho'}{Dt^2}\end{aligned}$$

PRESSURE-STRAIN RATE CORRELATION

- Solution for fluctuating pressure field from convective wave equation

$$p'(\mathbf{x}, t) = \int_0^t dt' \int d^3\mathbf{x}' G(\mathbf{x} - \mathbf{x}', t - t') f(\mathbf{x}', t')$$

$$\hat{p}(\mathbf{k}, t) = \int_0^t dt' \hat{G}(\mathbf{k}, t - t') \hat{f}(\mathbf{k}, t')$$

where $f(\mathbf{x}', t')$ is right side of wave equation

$$\Pi_{ij}(\mathbf{r}, \mathbf{t}) = \left\langle p'(\mathbf{x} + \mathbf{r}, \mathbf{t}) \left(\frac{\partial}{\partial x_j} u_i''(\mathbf{x}, \mathbf{t}) + \frac{\partial}{\partial x_i} u_j''(\mathbf{x}, \mathbf{t}) \right) \right\rangle$$

PRESSURE-STRAIN RATE CORRELATION

- Pressure-strain rate correlation from convective wave equation

$$\begin{aligned} \Pi_{ij}^r &= 2\bar{\rho} \left(\frac{M_g}{M_t} \right) \int_0^\infty dk \int_0^\pi \sin \theta d\theta \int_0^{2\pi} d\phi R(k, \theta, \phi; \tau_I) \\ &\quad \times k_l \tilde{S}_{ln} [E_{ni}(k, \theta, \phi; b_{mn})k_j + E_{nj}(k, \theta, \phi; b_{mn})k_i] \end{aligned}$$

$$\begin{aligned} E_{ij}(\mathbf{k}; \mathbf{b}) &= \frac{E(k)}{4\pi k^2} \left(\delta_{ij} - \hat{k}_i \hat{k}_j \right) + \frac{E_a(k)}{8\pi k^2} \left(\hat{k}_n b_{nm} \hat{k}_m \right) \left(\delta_{ij} + \hat{k}_i \hat{k}_j \right) \\ &\quad + \frac{E_a(k)}{4\pi k^2} \left[b_{ij} - \left(b_{in} \hat{k}_n \hat{k}_j + \hat{k}_i \hat{k}_n b_{nj} \right) \right] \end{aligned}$$

SOLENOIDAL DISSIPATION RATE BUDGET

● $\bar{\rho}\epsilon \approx \bar{\rho}\epsilon + \bar{\rho}\epsilon_d + \bar{\rho}\epsilon_I$

$$\bar{\rho}\epsilon = \overline{\bar{\mu}w'_{ij}w'_{ij}} = \overline{\bar{\mu}\omega'_i\omega'_i}$$

$$\bar{\rho}\epsilon_d = \frac{4}{3}\overline{\bar{\mu}s'_{kk}s'_{ll}}$$

$$\bar{\rho}\epsilon_I = 2\bar{\mu} \left[\frac{\partial^2(\overline{u'_i u'_j})}{\partial x_i \partial x_j} - 2 \frac{\partial}{\partial x_k} \left(\overline{u'_k \frac{\partial u'_j}{\partial x_j}} \right) \right]$$

- Contributions involving fluctuating viscosity negligible

SOLENOIDAL DISSIPATION RATE BUDGET

$$-A_\varepsilon + P_\varepsilon^1 + P_\varepsilon^2 + P_\varepsilon^3 + T_\varepsilon + D_\varepsilon - \Upsilon + B_\varepsilon + F_\varepsilon + \frac{\varepsilon}{\bar{\nu}} \frac{D\bar{\nu}}{Dt} = 0$$

$$P_\varepsilon^1 = 2\bar{\nu} \left[\left(\overline{\omega'_i \omega'_k} \bar{S}_{ik} - \overline{\omega'_i \omega'_i} \bar{S}_{kk} \right) + \left(\overline{\omega'_i s'_{ik}} \bar{\Omega}_k - \overline{\omega'_i s'_{kk}} \bar{\Omega}_i \right) \right]$$

Production mean shear, dilatation, vorticity

$$P_\varepsilon^2 = -2\bar{\nu} \left(\overline{u'_k \omega'_i} \right) \frac{\partial \bar{\Omega}_i}{\partial x_k} \quad \text{Gradient production}$$

$$P_\varepsilon^3 = \bar{\nu} \left(2\overline{\omega'_i \omega'_j s'_{ji}} - \overline{\omega'_i \omega'_i s'_{kk}} \right) \quad \text{Production due to vortex stretching}$$

$$T_\varepsilon = -\bar{\nu} \frac{\partial (\overline{u'_k \omega'_i \omega'_i})}{\partial x_k} \quad \text{Turbulent transport}$$

$$D_\varepsilon = 2\bar{\nu} \frac{\partial}{\partial x_k} \left[\frac{\omega'_r}{\rho} \left(e_{rji} \frac{\partial \sigma_{ik}}{\partial x_j} \right) \right] \quad \text{Viscous diffusion}$$

SOLENOIDAL DISSIPATION RATE BUDGET

$$-A_\varepsilon + P_\varepsilon^1 + P_\varepsilon^2 + P_\varepsilon^3 + T_\varepsilon + D_\varepsilon - \Upsilon + B_\varepsilon + F_\varepsilon + \frac{\varepsilon}{\bar{\nu}} \frac{D\bar{\nu}}{Dt} = 0$$

$$\Upsilon = 2\bar{\nu} \overline{\left[\frac{\partial}{\partial x_k} \left(\frac{\omega'_r}{\rho} \right) \right] \left(e_{rji} \frac{\partial \sigma_{ik}}{\partial x_j} \right)}, \quad \text{Viscous destruction}$$

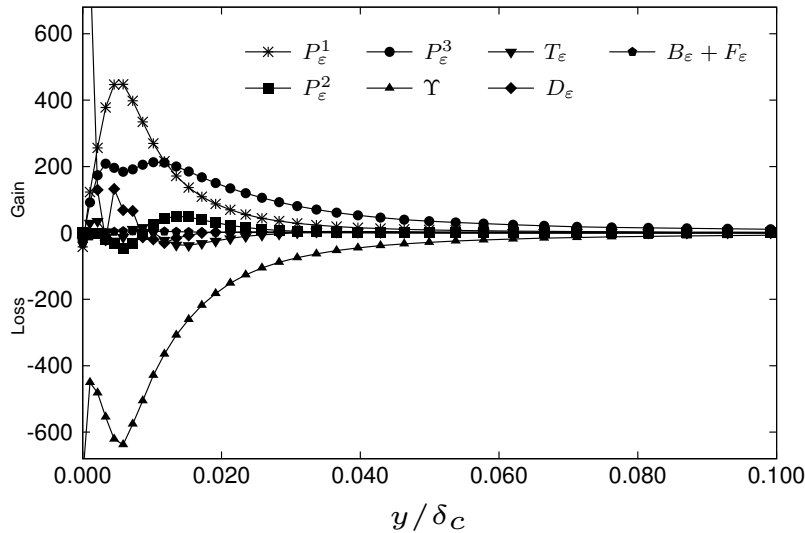
- Compressibility terms

$$B_\varepsilon = -2\bar{\nu} \overline{\left(\frac{\omega'_r}{\rho^2} \right) \left[e_{rij} \frac{\partial \rho}{\partial x_j} \frac{\partial p}{\partial x_i} \right]} \quad \text{Baroclinic term}$$

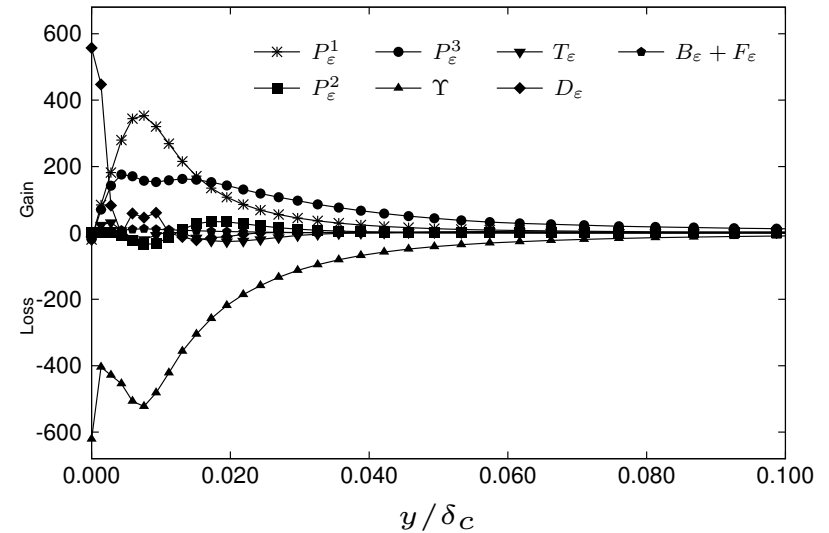
$$F_\varepsilon = 2\bar{\nu} \overline{\left(\frac{\omega'_r}{\rho^2} \right) \left[e_{rji} \frac{\partial \rho}{\partial x_j} \frac{\partial \sigma_{ik}}{\partial x_k} \right]}. \quad \text{Viscous force term}$$

- Last term on right known

SOLENOIDAL DISSIPATION RATE BUDGET



Isothermal



Adiabatic

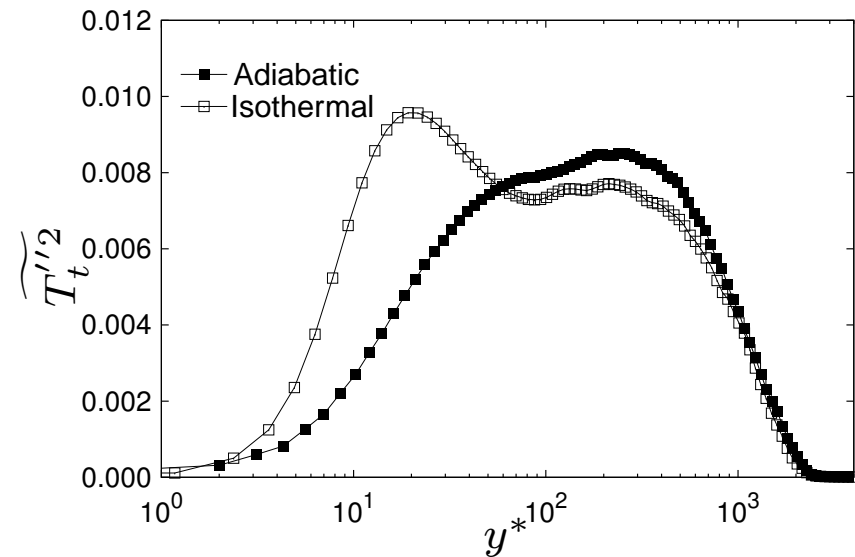
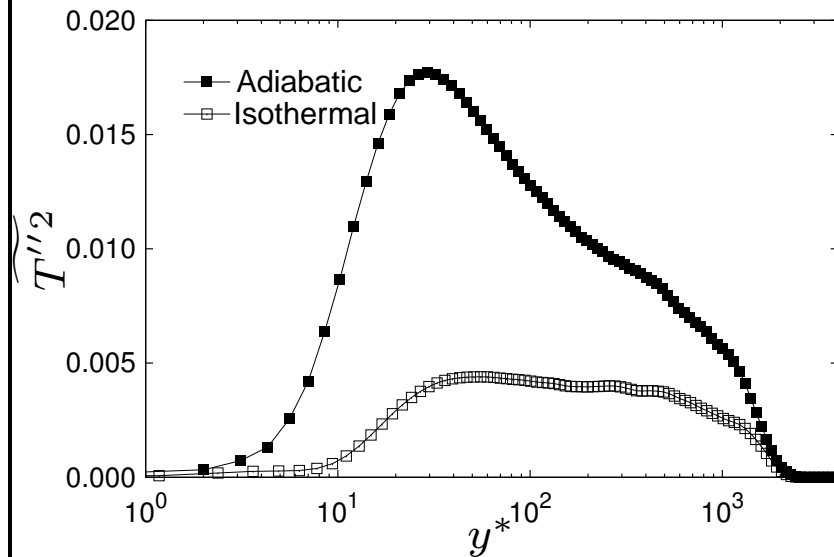
$$P_\epsilon^1 = 2\bar{\nu} \left[\left(\overline{\omega'_i \omega'_k} \bar{S}_{ik} - \overline{\omega'_i \omega'_i} \bar{S}_{kk} \right) + \left(\overline{\omega'_i s'_{ik}} \bar{\Omega}_k - \overline{\omega'_i s'_{kk}} \bar{\Omega}_i \right) \right]$$

$$P_\epsilon^3 = \bar{\nu} \left(2\overline{\omega'_i \omega'_j s'_{ji}} - \overline{\omega'_i \omega'_i s'_{kk}} \right)$$

$$\Upsilon = 2\bar{\nu} \left[\frac{\partial}{\partial x_k} \left(\frac{\omega'_r}{\rho} \right) \right] \left(e_{rji} \frac{\partial \sigma_{ik}}{\partial x_j} \right)$$

P_ϵ^3 relatively unaffected by change in wall condition

DISTRIBUTION OF TEMPERATURE (TOTAL) VARIANCES



- Temperature variance
 - Reduction of temperature fluctuations
 - Relatively constant over log-layer region
- Total temperature variance
 - Overshoot in buffer layer
 - Relatively constant and lesser in amplitude log-layer region

TEMPERATURE VARIANCE BUDGET

$$-A_T + P_T + T_T + D_T - \varepsilon_T + C_T^1 + C_T^2 = 0 ,$$

$$P_T = -\overline{\rho u_k'' \widetilde{T''}} \frac{\partial \widetilde{T}}{\partial x_k} \quad \text{Thermal production}$$

$$T_T = \frac{\partial}{\partial x_k} \left(\frac{\overline{\rho u_k'' T''^2}}{2} \right) \quad \text{Thermal transport}$$

$$D_T = \left(\frac{\gamma}{c_p} \right) \frac{\partial}{\partial x_k} \left[\overline{k_T T'' \frac{\partial T'}{\partial x_k}} + \overline{k_T' T'' \frac{\partial \overline{T}}{\partial x_k}} + \overline{k_T' T'' \frac{\partial T'}{\partial x_k}} \right] \quad \text{Thermal diffusion}$$

$$\varepsilon_T = \left(\frac{\gamma}{c_p} \right) \left[\overline{k_T \frac{\partial T''}{\partial x_k} \frac{\partial T'}{\partial x_k}} + \overline{k_T' \frac{\partial T''}{\partial x_k} \frac{\partial \overline{T}}{\partial x_k}} + \overline{k_T' \frac{\partial T''}{\partial x_k} \frac{\partial T'}{\partial x_k}} \right]$$

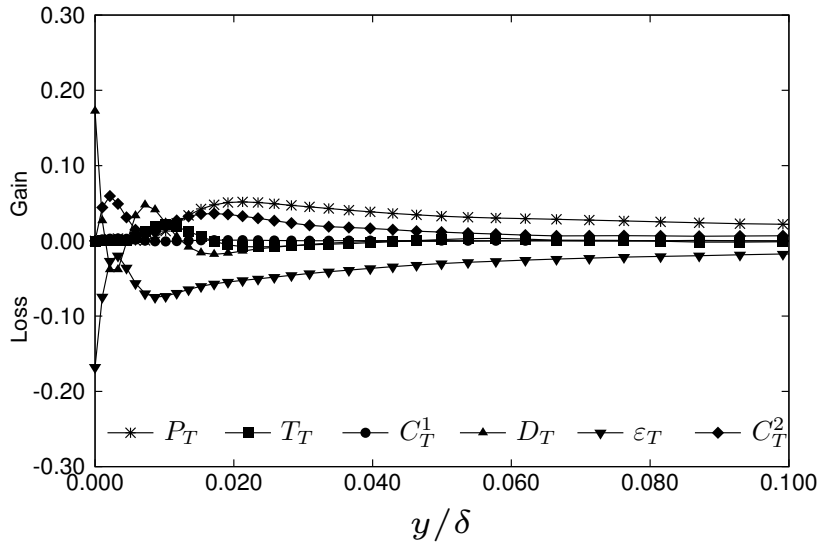
Thermal dissipation rate

$$C_T^1 = \left(\frac{\gamma}{c_p} \right) \left[\overline{T'' \frac{\partial}{\partial x_k} \left(\overline{k_T \frac{\partial \overline{T}}{\partial x_k}} \right)} \right], \quad C_T^2 = - \left(\frac{\gamma}{c_p} \right) \left[\overline{T'' \left(p \frac{\partial u_k}{\partial x_k} - \sigma_{ik} \frac{\partial u_i}{\partial x_k} \right)} \right]$$

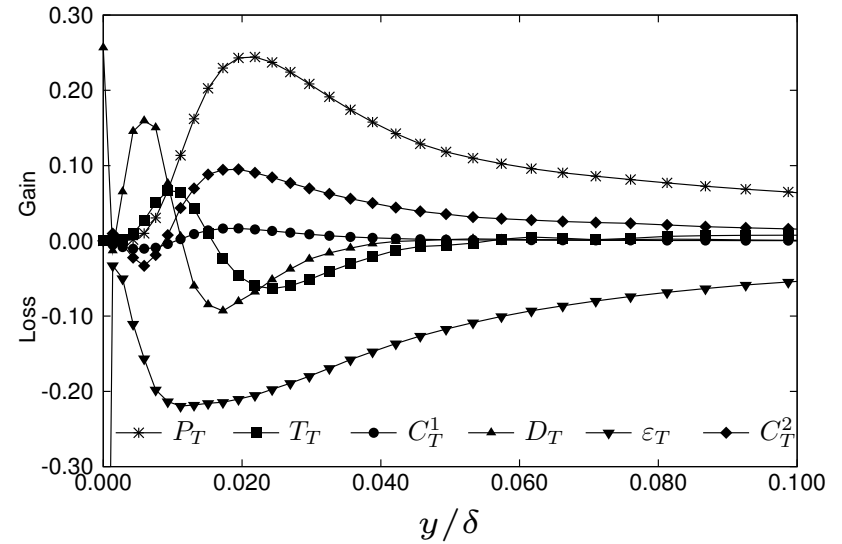
Thermal conduction

Pressure-dilatation, viscous dissipation

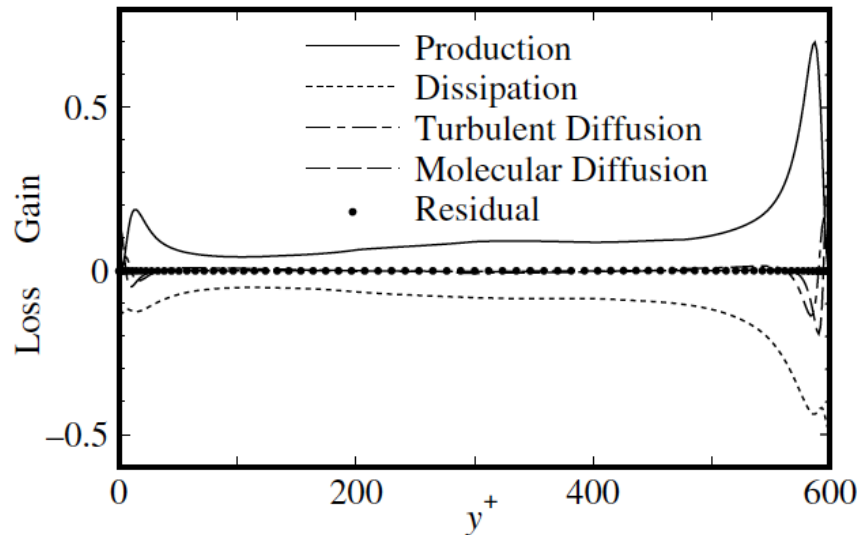
TEMPERATURE VARIANCE BUDGET



Isothermal



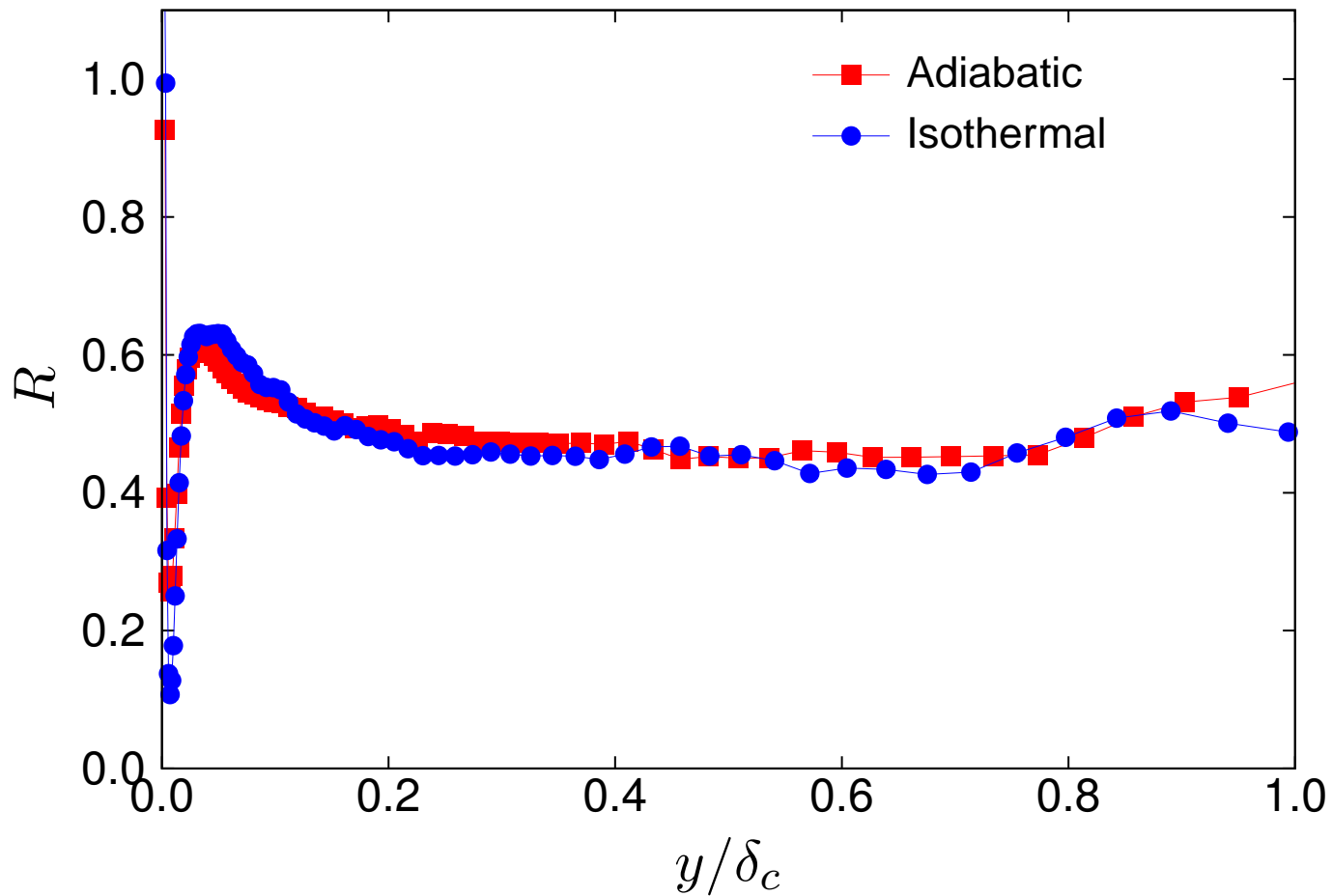
Adiabatic



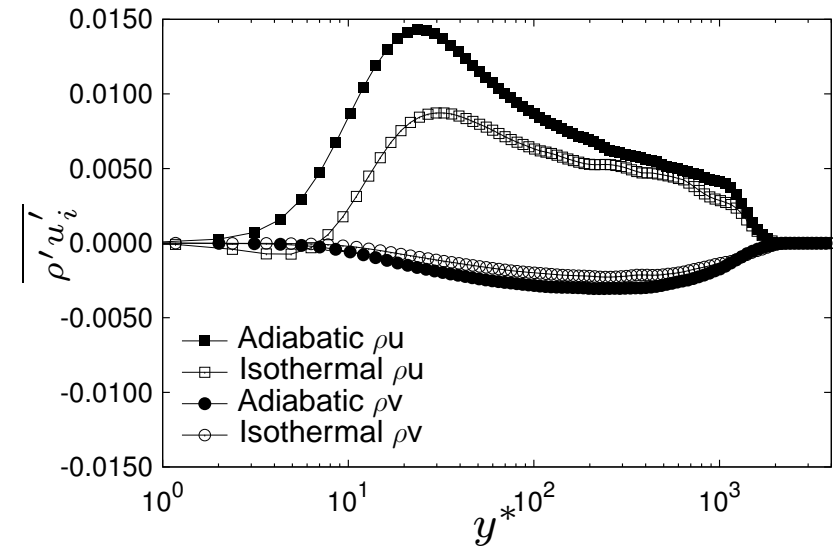
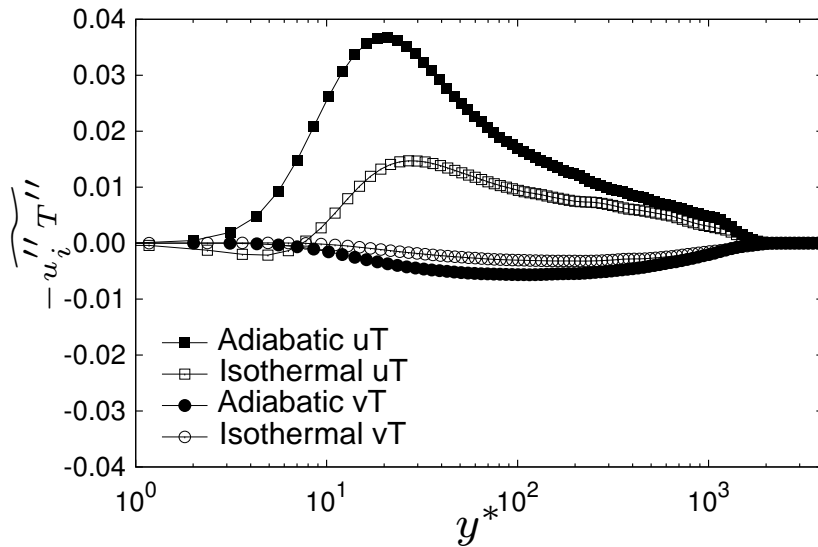
Morinishi et al. IJHMT 2007 ($T_w > T_{ad}$)

TIME SCALE RATIO

$$R = \left(\frac{K_T}{\varepsilon_T} \right) / \left(\frac{K}{\varepsilon} \right)$$



DISTRIBUTION OF HEAT AND MASS FLUX



$$\frac{\overline{p'u'_i}}{\bar{p}} = \frac{\widetilde{u''_i T''}}{\widetilde{T}} + \frac{\overline{\rho'u'_i}}{\bar{\rho}} .$$

● Heat and mass flux

- Reduction of streamwise heat flux component
- Minimal effect on $-\widetilde{v'' T''}$ and $\overline{\rho'v'}$ components

HEAT FLUX BUDGETS

$$-A_{Ti} + P_{Ti}^1 + P_{Ti}^2 + \Phi_{Ti} + D_{Ti} - \varepsilon_{Ti} + C_{Ti} = 0 ,$$

the production due to the mean temperature and velocity gradients,

$$P_{Ti}^1 = -\bar{\rho}\tau_{ik} \frac{\partial \tilde{T}}{\partial x_k} - \bar{\rho} \widetilde{u_k'' T''} \frac{\partial \tilde{u}_i}{\partial x_k} - \left(\frac{\gamma}{c_p} \right) \left[\overline{p u_i'' \tilde{S}_{kk}} - \overline{u_i'' \sigma'_{jk} \tilde{S}_{kj}} \right]$$

Production mean temperature, velocity gradients

$$P_{Ti}^2 = \left(\frac{\gamma}{c_p} \right) \left[\overline{u_i'' \sigma'_{jk} s''_{kj}} - \overline{u_i'' p' s''_{kk}} \right] \quad \text{Production fluctuating strain rate}$$

$$\Phi_{Ti} = - \left[T'' \frac{\partial p}{\partial x_i} \right] \quad \text{Pressure - scrambling}$$

HEAT FLUX BUDGETS

$$-A_{Ti} + P_{Ti}^1 + P_{Ti}^2 + \Phi_{Ti} + D_{Ti} - \varepsilon_{Ti} + C_{Ti} = 0 ,$$

$$D_{Ti} = \left(\frac{\gamma}{c_p} \right) \frac{\partial}{\partial x_k} \left[\overline{\bar{k}_T u_i'' \frac{\partial T'}{\partial x_k}} + \overline{k'_T u_i'' \frac{\partial \bar{T}}{\partial x_k}} + \overline{k'_T u_i'' \frac{\partial T'}{\partial x_k}} \right] \\ - \frac{\partial}{\partial x_k} \left[\overline{\rho u_i'' T'' u_k''} - \overline{\sigma'_{ik} T''} \right]$$

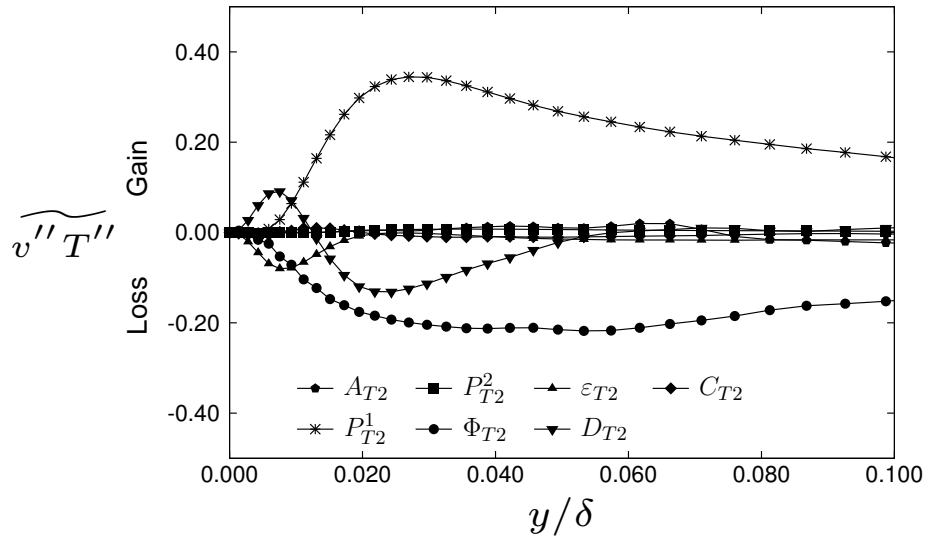
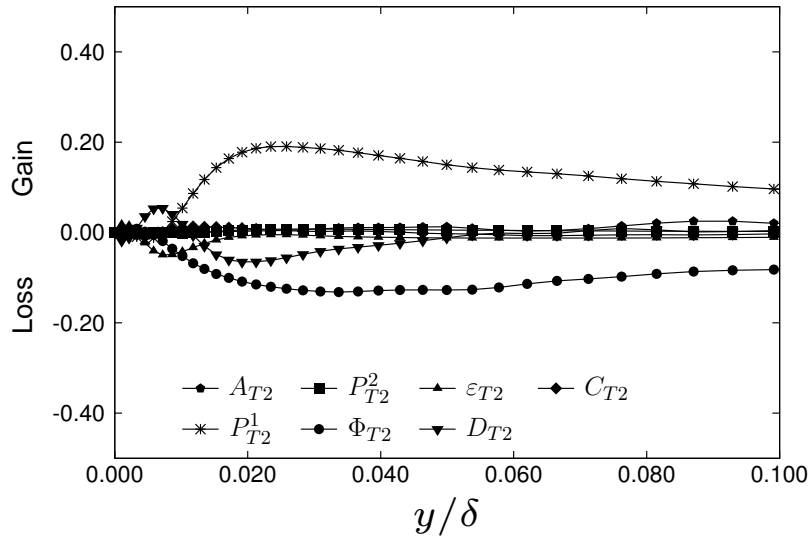
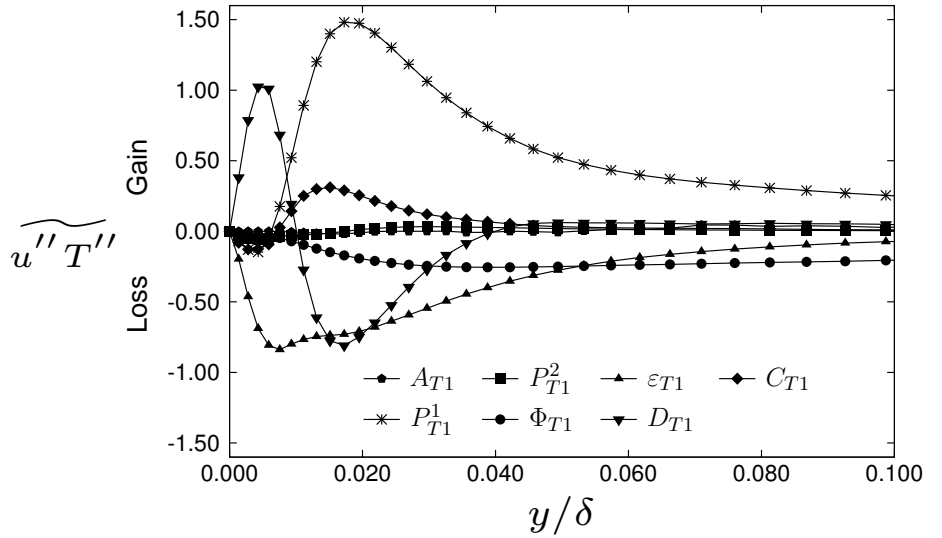
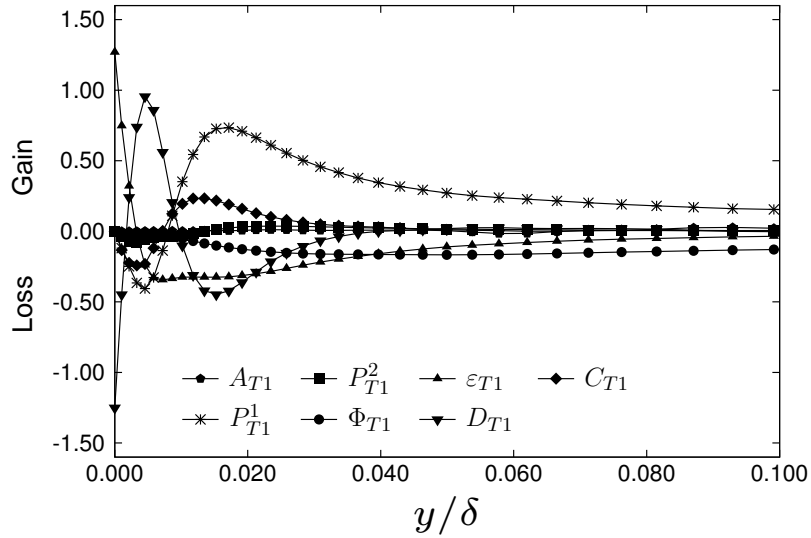
Turbulent, viscous – thermal transport

$$\varepsilon_{Ti} = \left(\frac{\gamma}{c_p} \right) \left[\overline{\bar{k}_T \frac{\partial u_i''}{\partial x_k} \frac{\partial T'}{\partial x_k}} + \overline{k'_T \frac{\partial u_i''}{\partial x_k} \frac{\partial \bar{T}}{\partial x_k}} + \overline{k'_T \frac{\partial u_i''}{\partial x_k} \frac{\partial T'}{\partial x_k}} \right] + \left[\overline{\sigma'_{ik} \frac{\partial T''}{\partial x_k}} \right]$$

Turbulent, viscous – thermal dissipation

$$C_{Ti} = \overline{T''} \frac{\partial \bar{\sigma}_{ik}}{\partial x_k} + \left(\frac{\gamma}{c_p} \right) \left[\overline{u_i'' \bar{\sigma}_{jk} \tilde{S}_{kj}} + \overline{u_i'' s''_{kj}} (\bar{\sigma}_{kj} - \bar{p} \delta_{kj}) \right] \quad \text{Compressibility}$$

HEAT FLUX BUDGETS



Isothermal

Adiabatic

RANS MODEL DEVELOPMENT

- Higher-Order Velocity Correlation Models
 - Pressure-strain rate, pressure-dilation and solenoidal dissipation
 - Variable density extensions for other unknown correlations
 - Wall-proximity modifications
- Temperature Variance Models
 - Variance dissipation rate model
 - Numerous unknown correlations
 - Wall-proximity modifications
- Thermal-flux Models
 - Pressure-scrambling term important
 - Numerous unknown correlations
 - Wall-proximity modifications

RANS MODEL DEVELOPMENT

- Low-Order Velocity Correlation Models
 - Wide variety of two-equation models
 - Heat flux vector
 - Gradient transport models - wall proximity corrections
 - Variable Pr_t effects
 - Variable time scale ratio
- Hybrid Methods
 - Simplest level - provides RANS solution where LES is deficient
 - Complex level - provides linkage between DNS/LES and RANS

References

- Barone, M. F., Oberkampf, W. L., Blottner, F. G., 2006. Validation case study: Prediction of compressible turbulent mixing layer growth rate. *AIAA J.* 44, 1488–1497.
- Birch, S. F., Eggers, J. M., 1972. A critical review of the experimental data for developed free turbulent shear flows. In: *Free Turbulent Shear Flows*. No. 321 in NASA. pp. 11–40.